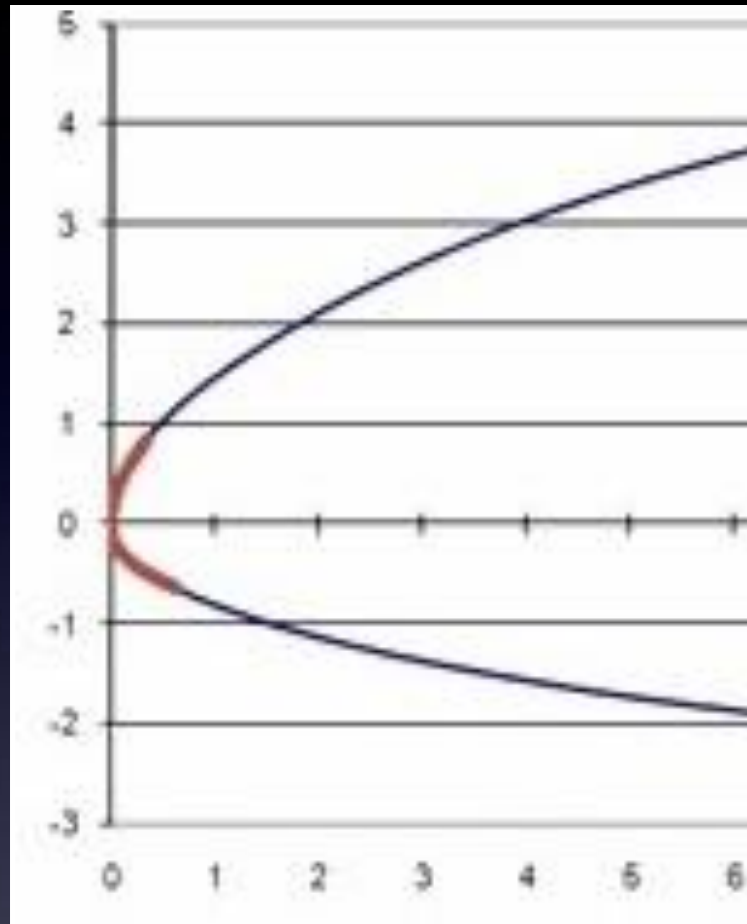


# Leading Edge Tape

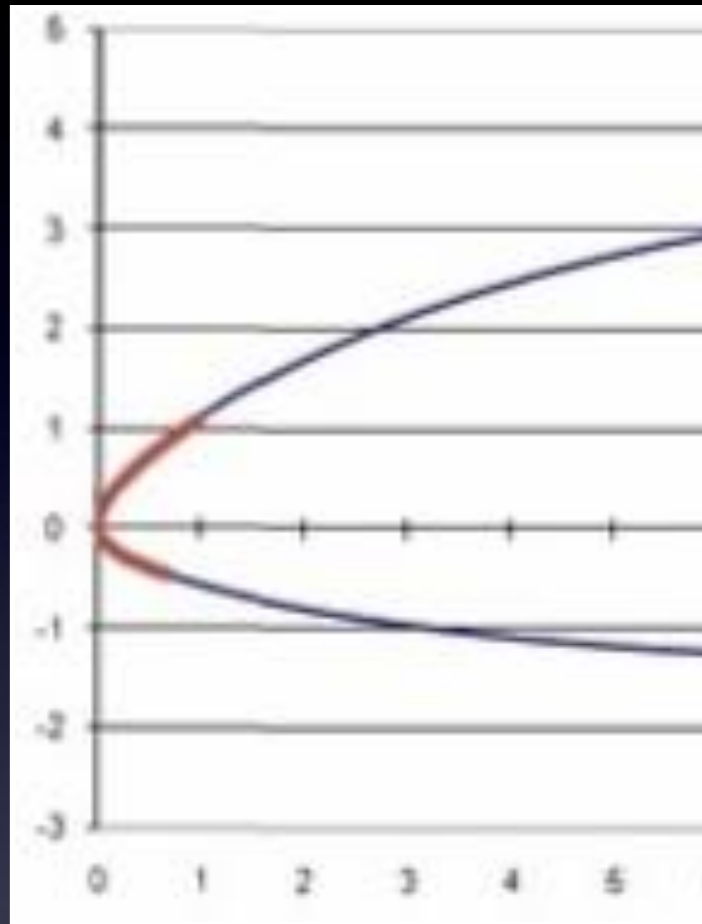
## Humidity Dependence

Jim Hendrix  
The Deturbulator Project  
[www.deturbulator.org](http://www.deturbulator.org)

# Position of Tape



**Wing Root**



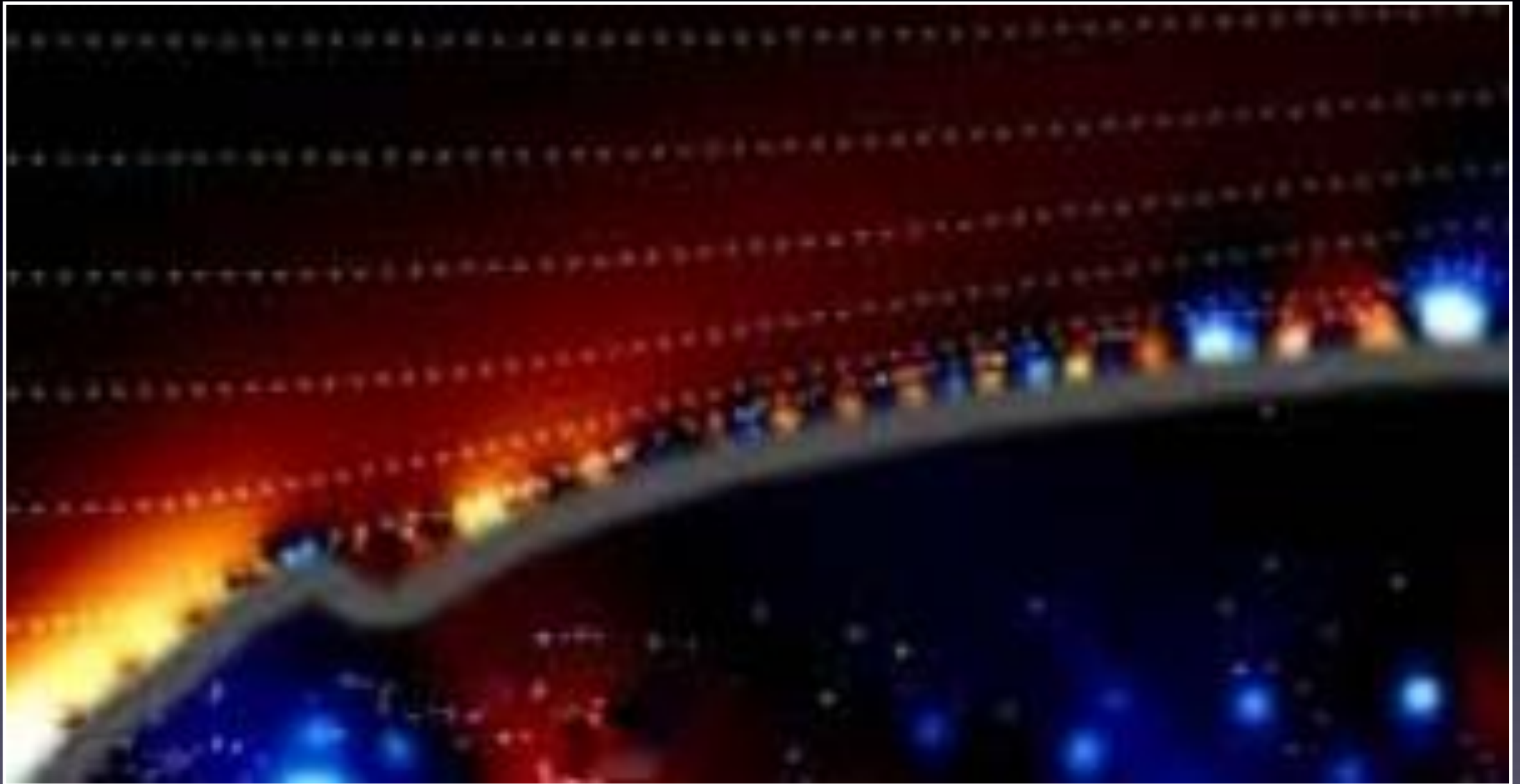
**Inboard End of Aileron**



**Wing Tip**

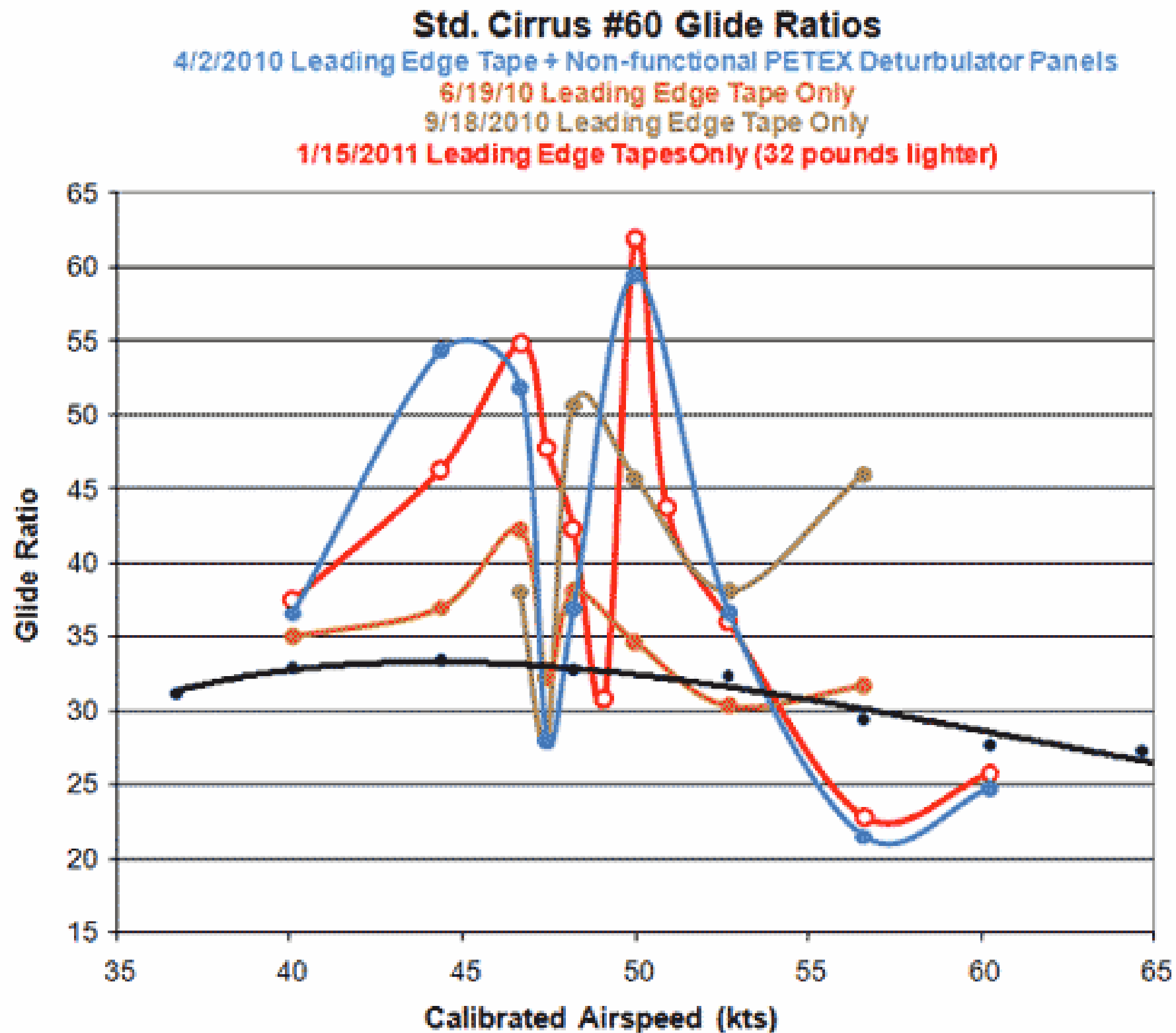
**Tape Thickness: .0025"**

# Demonstration of Hypothetical Vortical Separation Layer

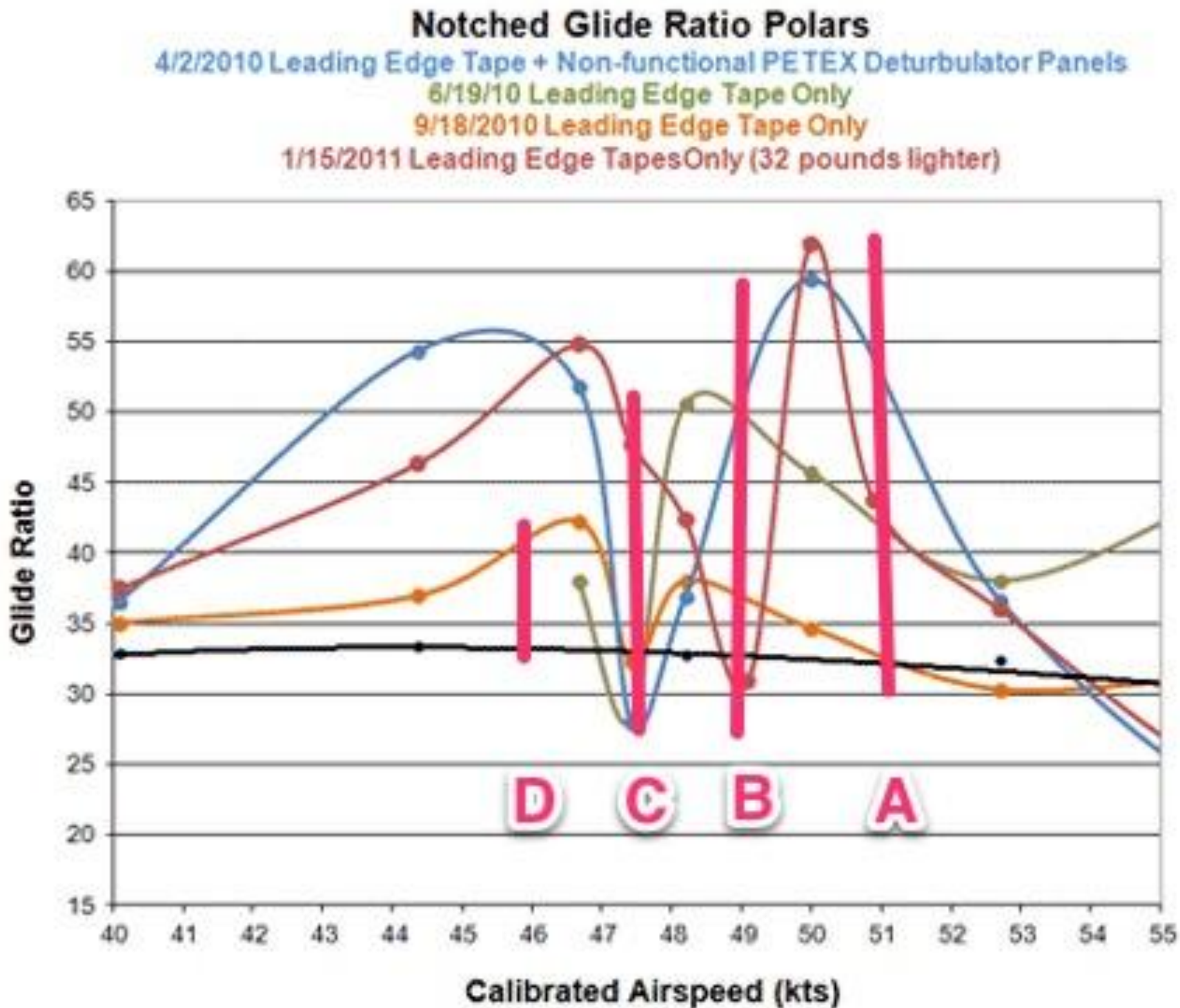


This is not a realistic simulation. It is only a demonstration of the sort of vortical flow that is postulated.

# Four Glide-Ratio Measurements

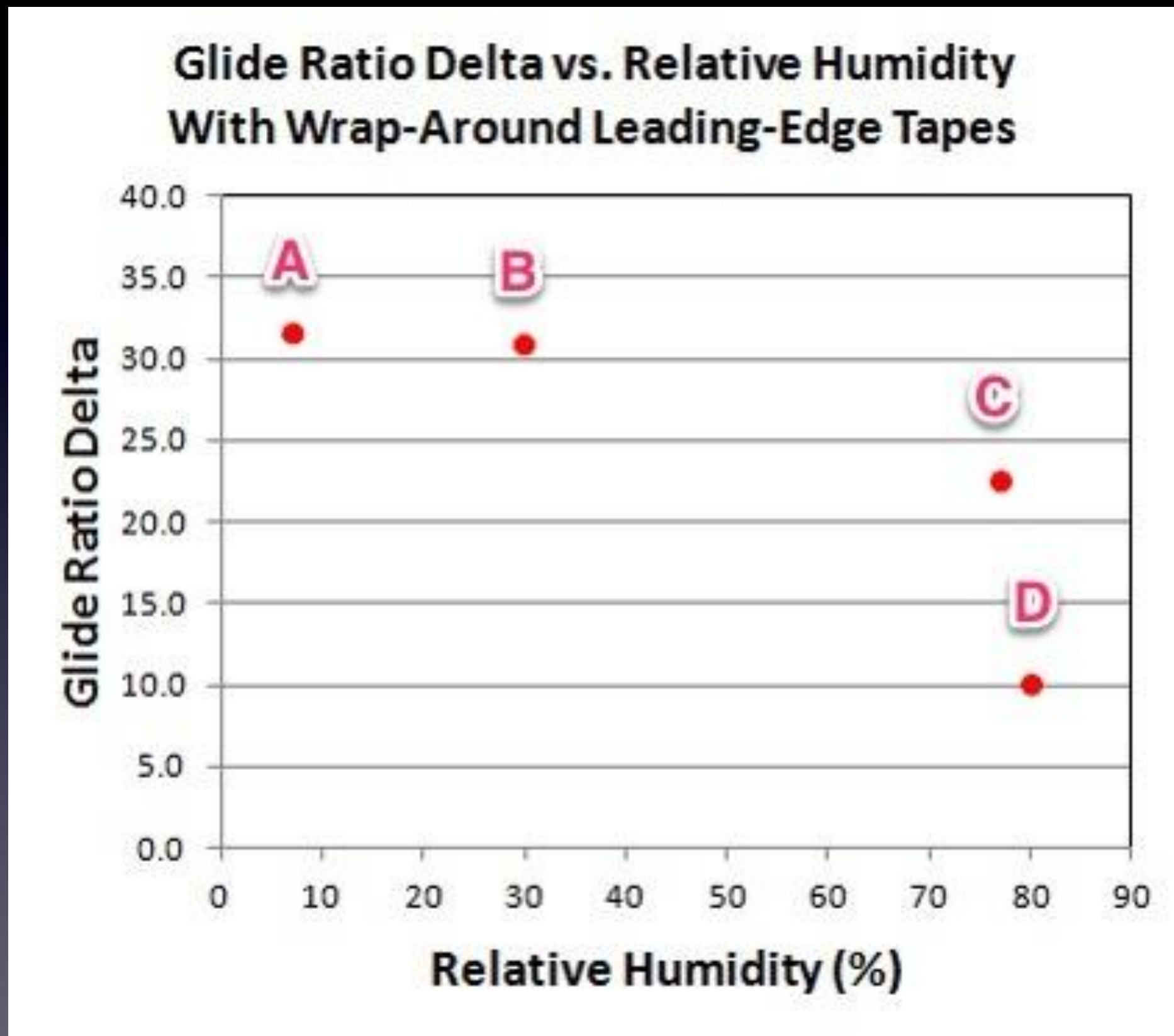


# Amplitude Deltas

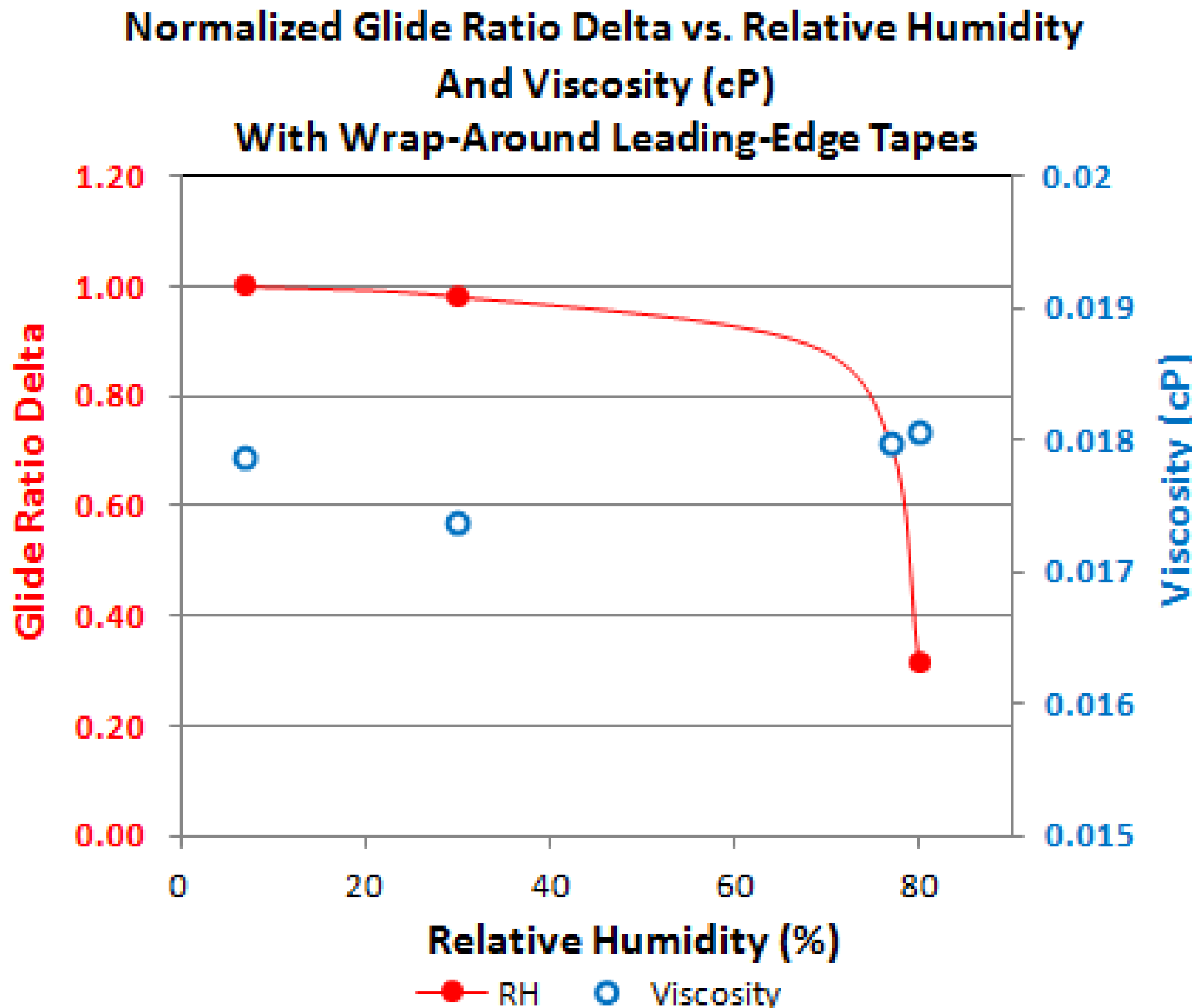




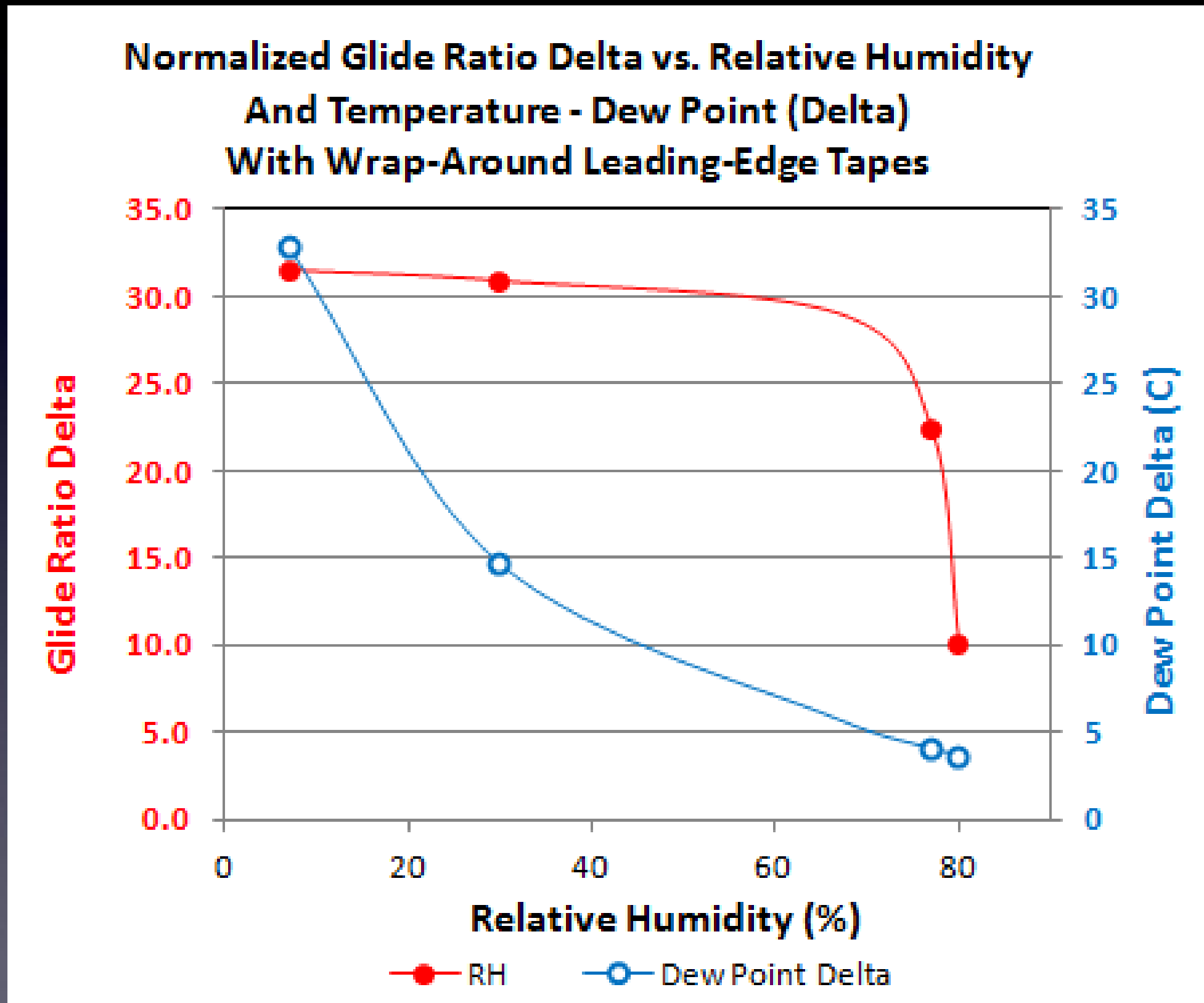
# Humidity Dependency



# No Viscosity Correlation



# (Temperature - DewPoint) Correlation





# Observations

- The large swings resulting from an upper surface rear-facing step diminish sharply when relative humidity exceeds 75%.
- This is thought to be due to a failure to generate a thin (.0025") bed of rolling vortexes that maintains detached flow.
- The tiny vortices require large surface friction values on the wing surface behind the tape.
- Viscosity does not correlate with performance amplitude deltas, but proximity of air temperature and dew point temperature does.
- The performance delta falls off sharply when the air temperature is within 4 degrees F of the dew point temperature...near saturation.
- Since viscosity does not correlate, this implies a skin friction loss.

**End**